

## Appendix - A1

### NON CIRCULAR BEARING GEOMETRY

The non circular (two-lobe, three lobe) bearing geometries are as follows.

#### A 1.1 Two-Lobe Bearing

The two lobe bearing consists of two circular halves with centers  $O_1$  and  $O_2$  (Fig. A 1.1) with offset  $\varepsilon_p$  on either side of the geometric centre  $O_b$  of the bearing. The eccentricities  $\varepsilon_1$  and  $\varepsilon_2$  and attitude angles  $\phi_1$  and  $\phi_2$  for any journal centre  $O_3$  can be calculated by using the following equations.

For lower lobe:

$$\varepsilon_1 = \left\{ \varepsilon^2 + \varepsilon_p^2 + 2\varepsilon\varepsilon_p \cos \phi \right\}^{1/2} \quad (\text{A 1.1})$$

$$\phi_1 = \tan^{-1} \left\{ \frac{\varepsilon \sin \phi}{\varepsilon_p + \varepsilon \cos \phi} \right\} \quad (\text{A 1.2})$$

For upper lobe:

$$\varepsilon_1 = \left\{ \varepsilon^2 + \varepsilon_p^2 - 2\varepsilon\varepsilon_p \cos \phi \right\}^{1/2} \quad (\text{A 1.3})$$

$$\phi_2 = \pi - \tan^{-1} \left\{ \frac{\varepsilon \sin \phi}{\varepsilon_p + \varepsilon \cos \phi} \right\} \quad (\text{A 1.4})$$

#### A 1.2 Three-Lobe Bearing

The three lobe bearing consists of three non concentric arcs Fig. A 1.2 and the various geometric parameters of a three lobe bearing are related as follows.

Lobe 1:

$$\varepsilon_1 = \left\{ \varepsilon^2 + \varepsilon_p^2 + 2\varepsilon\varepsilon_p \cos \phi \right\}^{1/2} \quad (\text{A 1.5})$$

$$\phi_1 = \tan^{-1} \left\{ \frac{\varepsilon \sin \phi}{\varepsilon_p + \varepsilon \cos \phi} \right\} \quad (\text{A 1.6})$$

Lobe 2:

$$\varepsilon_1 = \left\{ \varepsilon^2 + \varepsilon_p^2 - 2\varepsilon\varepsilon_p \cos\left(\frac{\pi}{3} + \phi\right) \right\}^{1/2} \quad (\text{A 1.7})$$

$$\phi_2 = \frac{2\pi}{3} - \tan^{-1} \left\{ \frac{\varepsilon \sin\left(\frac{\pi}{3} + \phi\right)}{\varepsilon_p - \varepsilon \cos\left(\frac{\pi}{3} + \phi\right)} \right\} \quad (\text{A 1.8})$$

Lobe 3:

$$\varepsilon_1 = \left\{ \varepsilon^2 + \varepsilon_p^2 - 2\varepsilon\varepsilon_p \cos\left(\frac{\pi}{3} - \phi\right) \right\}^{1/2} \quad (\text{A 1.9})$$

$$\phi_2 = \frac{2\pi}{3} - \tan^{-1} \left\{ \frac{\varepsilon \sin\left(\frac{\pi}{3} - \phi\right)}{\varepsilon_p - \varepsilon \cos\left(\frac{\pi}{3} - \phi\right)} \right\} \quad (\text{A 1.10})$$

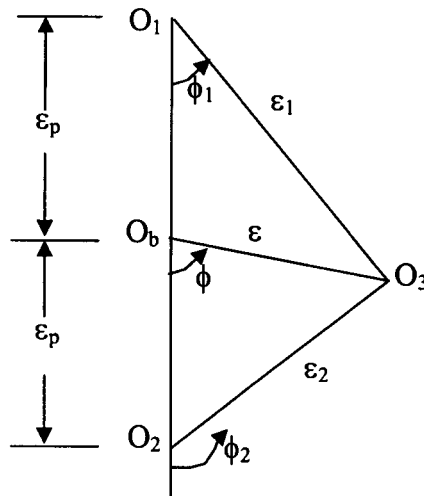


Fig. A 1.1 Geometry of two lobe bearing

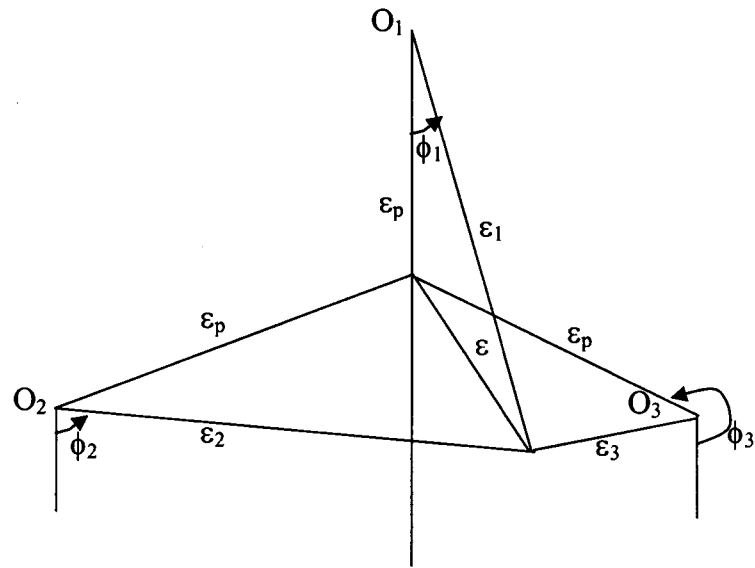


Fig. A 1.2 Geometry of three lobe bearing

## Appendix-A2

### THE TRANSFORMATION OF STIFFNESS AND DAMPING COEFFICIENTS

The fluid film damping forces which depend on velocities  $\dot{\xi}$  and  $\dot{\eta}$  are given by its horizontal and vertical components and these components can be written as shown below.

$$W_1 = -(b_{1\xi} \dot{\xi} + b_{1\eta} \dot{\eta}) \quad (\text{A 2.1})$$

$$W_2 = -(b_{2\xi} \dot{\xi} + b_{2\eta} \dot{\eta}) \quad (\text{A 2.2})$$

Equations A 2.1 and A 2.2 can be written in matrix form as

$$\begin{bmatrix} W_1 \\ W_2 \end{bmatrix} = \begin{bmatrix} b_{1\xi} / \varepsilon & b_{1\eta} \\ b_{2\xi} / \varepsilon & b_{2\eta} \end{bmatrix} \begin{bmatrix} \varepsilon \dot{\xi} \\ \dot{\eta} \end{bmatrix} \quad (\text{A 2.3})$$

$\varepsilon$ ,  $\dot{\xi}$  and  $\dot{\eta}$  can be written in terms of  $\dot{x}_1$  and  $\dot{x}_2$  as given below.

$$\begin{bmatrix} \varepsilon \dot{\xi} \\ \dot{\eta} \end{bmatrix} = \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix} \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} \quad (\text{A 2.4})$$

Thus equation A 2.3 can be written as

$$\begin{bmatrix} W_1 \\ W_2 \end{bmatrix} = \begin{bmatrix} b_{1\xi} / \varepsilon & b_{1\eta} \\ b_{2\xi} / \varepsilon & b_{2\eta} \end{bmatrix} \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix} \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} \quad (\text{A 2.5})$$

The fluid film damping force components can be written in terms of damping coefficients and velocities as given below.

$$\begin{bmatrix} W_1 \\ W_2 \end{bmatrix} = \begin{bmatrix} B_{11} & B_{12} \\ B_{21} & B_{22} \end{bmatrix} \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} \quad (\text{A 2.6})$$

Comparing equations A 2.5 and 2.6

$$\begin{bmatrix} B_{11} & B_{12} \\ B_{21} & B_{22} \end{bmatrix} = \begin{bmatrix} b_{1\xi} & b_{1\eta} \\ b_{2\xi} & b_{2\eta} \end{bmatrix} \begin{bmatrix} \frac{\cos\theta}{\varepsilon} & \frac{-\sin\theta}{\varepsilon} \\ \sin\theta & \cos\theta \end{bmatrix} \quad (\text{A 2.7})$$

As the displacements and velocities are in the same direction similar transformation can be used for stiffness coefficients also.

$$\begin{bmatrix} S_{11} & S_{12} \\ S_{21} & S_{22} \end{bmatrix} = \begin{bmatrix} s_{1\xi} & s_{1\eta} \\ s_{2\xi} & s_{2\eta} \end{bmatrix} \begin{bmatrix} \frac{\cos\theta}{\varepsilon} & \frac{-\sin\theta}{\varepsilon} \\ \sin\theta & \cos\theta \end{bmatrix} \quad (\text{A 2.8})$$

